

SHEET 8

Nyquist Diagram

[1] A unity feedback system has a forward transfer function: $G(s) = \frac{k}{s(s+1)(s+0.5)}$

- Sketch the Nyquist diagram for $k=1, 0.75$ and 0.5 .
- For the above cases, determine :
 - Whether the system is stable. Check using Routh method.
 - The gain crossover and the phase crossover frequencies.
 - The gain and phase margins.
- Adjust the gain such that the $GM=2\text{db}$.
- Adjust the gain such that the $PM=10$ degrees.

[2] A unity feedback system has a forward transfer function: $G(s) = \frac{(s+2)}{s^2(s+0.5)}$

- Sketch the Nyquist diagram.
- For the above cases, determine :
 - Whether the system is stable. Check using Routh method.
 - The gain crossover and the phase crossover frequencies.
 - The gain and phase margins.
- Adjust the gain such that the $GM=2\text{db}$.
- Adjust the gain such that the $PM=10$ degrees.

[3] (midterm2001/Final 2002) A system with unity feedback has the forward transfer function: $G(S) = \frac{K}{(S+1)(S^2+2S+2)}$

- Find the limiting value of K for stability.
- Sketch the Nyquist plot for $K = 2.5$ based on points calculated at $\omega = 0, 0.5, 1, 1.5, 1.8, 2$ and ∞ rad/sec.
- For the value of K obtained in (b), find the gain crossover frequency ω_{gc} , the phase crossover frequency ω_{pc} , the gain margin GM , the phase margin PM and steady-state error e_{ss} for a unit step input.
- Determine the new gain corresponding to a phase margin of 30° , and the new steady-state error $e_{ss\text{ new}}$ for a unit step input.

[4](Midtem97/2002) In a frequency response experiment, it was found that: Gain crossover frequency, $\omega_{gc}=1.2$ rad/sec and Phase Margin=22.92°.

- (a) Assume that the closed loop system is dominated by a second-order model, find the closed loop transfer function of this system. (Note : Prove any formula used)
- (b) Assume that the above system is unity feedback and insert $G^*(s)=k/(s+1)$ in the forward path. For a desired gain margin of 20 db.

Determine:

- The static error coefficient k.
- The corresponding steady state error for a unit ramp input.
- The new phase margin.

[5](Final2001)

(a) The following experimental results were obtained from an open loop frequency response of an automatic control system.

ω (rad/sec)	4	5	6	8	10
Gain	0.66	0.48	0.36	0.23	0.15
Phase (°)	-134	-143	-152	-167	-180

- (i) Plot the Nyquist diagram of the open loop T.F.
- (ii) Obtain the values of ω_{pc} , GM and PM.(You can extrapolate the Nyquist locus).

[6] (Final2000) For a unity feedback type 1 control system, the forward transfer function is given by $G_p(s) = \frac{K}{s(1+s)(1+0.1s)}$. Using Nyquist plot,

- (a) Obtain the values of gain crossover frequency ω_{gc} , phase crossover frequency ω_{pc} , gain margin GM and phase margin PM for the normalized plant.
- (b) Find the limiting value of K for the above closed loop system to be marginally stable. Verify your result using Routh's stability criterion.
- (c) Estimate the corresponding values of ζ , ω_n and the bandwidth.
(Prove any formula you use)

Summary

1] Main steps:

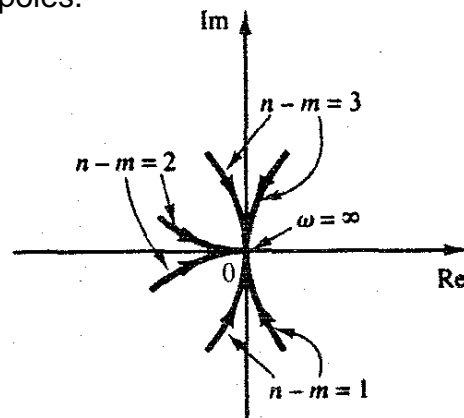
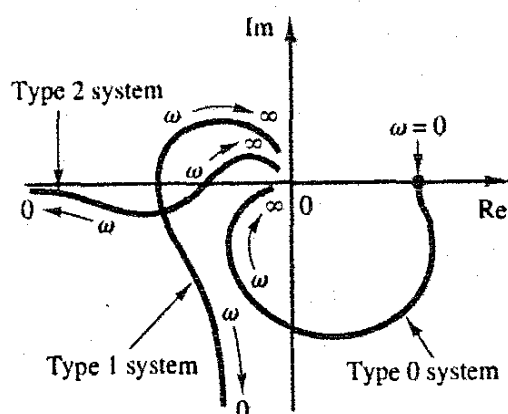
1. If you have a transfer function in the form of: $G(S) = \frac{K(S + z_1)(S + z_2) \dots (S + z_m)}{S^r(S + p_1)(S + p_2) \dots (S + p_n)}$
2. Rearrange the transfer function in the form of:

$$G(S) = \frac{K(1 + j\omega\tau_a)(1 + j\omega\tau_b) \dots (1 + j\omega\tau_m)}{S^r(1 + j\omega\tau_1)(1 + j\omega\tau_2) \dots (1 + j\omega\tau_n)}$$
3. determine the type of the system:
 - a. For type 0 system (r = 0): The starting point of the plot is finite and is on the positive real axis and equals K.
 - b. For type 1 system (r = 1): The starting point of the plot is infinite and is asymptotic to a line parallel to the negative imaginary axis.
 - c. For type 2 system (r = 2): The starting point of the plot is infinite and is asymptotic to a line parallel to the negative real axis.
4. get the magnitude and the phase of the transfer function in the form:

$$|G(S)| = \frac{K \sqrt{1 + (\omega\tau_a)^2} \sqrt{1 + (\omega\tau_b)^2} \dots}{\omega^r \sqrt{1 + (\omega\tau_1)^2} \sqrt{1 + (\omega\tau_2)^2} \dots}$$

$$\angle G(S) = -90r + \tan^{-1} \omega\tau_a + \tan^{-1} \omega\tau_b + \dots - \tan^{-1} \omega\tau_1 - \tan^{-1} \omega\tau_2 - \dots$$

5. substitute in the $|G(S)|$ and $\angle G(S)$ by:
 - a. All corner frequencies.
 - b. $\omega = 0$ and $\omega = \infty$.
 - c. You must have **3 points** at least in the third quarter and at least **one point** in the second quarter (if the plot exists in these regions).
6. At $\omega = \infty$ the plot will be tangent to one of the axes, according to the difference between the number of zeros and the number of poles.



II] Stability analysis using Nyquist diagram:

- $GM = \frac{1}{K_g}$ or $GM_{db} = -20 \log K_g$.

- $PM = 180 + \left| G(S) \right|_{\omega_{gc}} = 180 + \Phi$

- How to get ω_{gc} ?

1. $\left| G(S) \right|_{\omega_{gc}} = \Phi$

2. Take tan for the two sides.

(Remember $\tan(a \pm b) = \frac{\tan a + \tan b}{1 \mp \tan a \tan b}$)

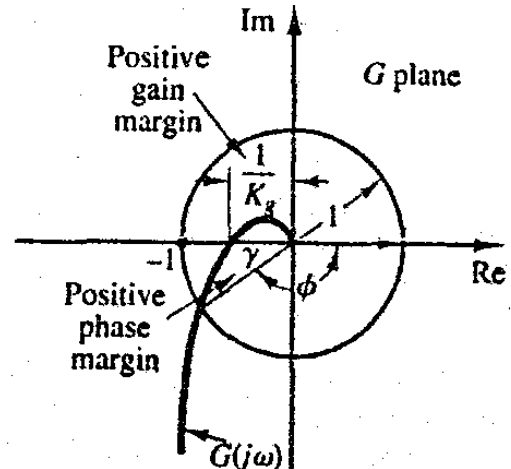
3. Solve to get ω_{gc} .

- How to get ω_{pc} ?

1. $\left| G(S) \right|_{\omega_{pc}} = -180$

2. Take tan for the two sides. (Remember $\tan(a \pm b) = \frac{\tan a + \tan b}{1 \mp \tan a \tan b}$)

3. Solve to get ω_{pc} .



III] System identification:

1. For a second order:

- If you are given the frequency response (ω_{gc} & PM) \rightarrow and required the time response (ζ & ω_n).

- $\frac{\omega_{gc}}{\omega_n} = \sqrt{\sqrt{1+4\zeta^4} - 2\zeta^2}$

- $PM = \tan^{-1} \frac{2\zeta\omega_n}{\omega_{gc}}$

- $\tan PM = \frac{2\zeta}{\sqrt{\sqrt{1+4\zeta^4} - 2\zeta^2}}$

- For $PM \leq 60^\circ \rightarrow \zeta \approx \frac{(PM)}{100}$.

- M_m or $M_r = \frac{1}{2\zeta\sqrt{1-\zeta^2}}$

- ω_m or $\omega_r = \omega_n \sqrt{1-\zeta^2}$

- $\omega_B = \sqrt{\omega_n^2(1-\zeta^2) + \omega_n \sqrt{2(1-\zeta^2\omega_n^2)}}$